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A Conservative Coupling Method for Fluid-Structure Interaction in the Compressible Case

Laurent Monasse, Virginie Daru, Christian Mariotti, and Serge Piperno

Abstract We propose a general explicit coupling method between a Finite Volume method for compressible flow and a rigid body. The coupling strategy is based on the idea of Embedded Boundary methods (Pember et al., J. Comput. Phys. 120:278–304, 1995). The fluids are computed everywhere in the Cartesian grid, and are modified at the solid boundaries to enforce fluid mass conservation. The coupling between the fluid and the solid is designed to ensure a balance in momentum and energy. We prove the exact numerical conservation of several simple uniform flows. An illustrative example of the liftoff of a cylinder by a shock wave is presented and compared with existing results.

1 Introduction

For fluid-structure interaction problems, two main types of methods were developed. The Arbitrary Lagrangian-Eulerian method (ALE) was first developed in the late seventies [4]. However, as a body-fitted method, the ALE technique requires remeshing of the fluid domain when the solid is subjected to large deformations or breaking, which can be computationally demanding. In order to avoid body-fitting and remeshing, Peskin [14] proposed the Immersed Boundary method for the coupling of incompressible biological fluid flows with moving elastic boundaries. Penalization type methods have been applied to compressible fluid-structure interaction [1], but the stiff condition on the time-step as the penalization parameter is increased is unsatisfactory. Direct forcing methods, which modify the values of the fluid cells in the vicinity of the solid boundary, have been widely applied [6, 10, 12], but still do not ensure the conservation of physical quantities at the interface.

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Conservative methods for Finite Volumes with complex geometries have been developed, such as the Embedded Boundary method [13]. Several procedures were proposed to avoid the stability condition restriction due to cut cells with small volume. We adopt here the solution consisting in merging the small cut-cells with their stable neighbors, that were successfully applied for compressible fluid-structure interaction [8].

In the sequel the equations for fluid modification remain similar to [8]. The main difference lies in the coupling approach. Ultimately, our aim is to couple the compressible flow with a Discrete Element method for the solid. As the Discrete Element method is computationally expensive, we use an explicit coupling method. In this paper, we restrict ourselves to rigid solids, but the method also applies for deformable bodies. The algorithm is carried out in a partitioned way: the fluid pressure makes the solid move, and the updated position of the solid induces a modification of the fluid fluxes at the new boundary. The fluid solver is based on the time splitting high order scheme developed in [3]. However the theoretical analysis of the coupling algorithm does not depend on the scheme that is used.

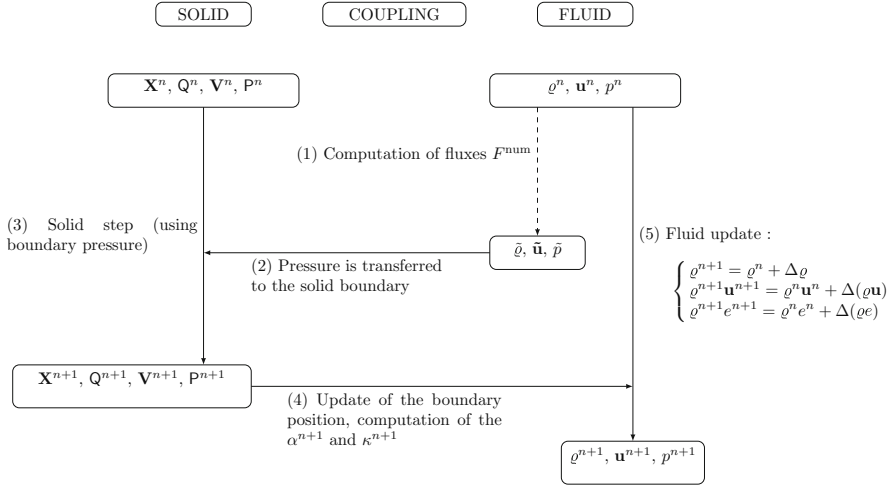
2 Description of the Method

The position of the solid in the fluid domain is taken into account using the Embedded Boundary method [5, 7–9, 13]. At time t , the solid occupies a volume fraction α_i of cell i , and all variables are assumed to be uniform in the cell. The conservative quantities contained in the cell are therefore equal to their value at the center of the cell times the volume of the cell and the volume fraction of fluid $1 - \alpha_i$. In the same way, the computed fluxes are assumed to be constant on cell faces. If we denote κ_{ij} the surface fraction of the face between cells i and j occupied by solid, we set the effective flux between i and j as the computed flux times the surface of their interface and the surface fraction of fluid $1 - \kappa_{ij}$. Additional fluxes come from the presence of the solid boundary, and are computed in order to yield exact conservation of fluid mass and of total momentum and energy of the system.

We now describe the algorithm we developed, which preserves the fully conservative properties of the Finite Volume method combined with the Embedded Boundary method. At the beginning of a time step, at time $n \Delta t$, the known quantities are:

- The position of the center of mass of the solid particle \mathbf{X}^n and the rotation matrix \mathbf{Q}^n
- The velocity of the center of mass of the solid particle \mathbf{V}^n and the angular quantity of movement matrix \mathbf{P}^n
- The fluid state: density ρ^n , velocity \mathbf{u}^n and pressure p^n .

The general algorithm is as follows:



The pressures p_x^{ref} and p_y^{ref} transferred to the solid boundary are chosen to be the temporal average of the reference pressures used during the partial steps of the fluid computation (1) in the x and y directions (in our case, due to the splitting method; these steps could also be the steps of a Runge-Kutta method). The solid is advanced in step (3) using a classical second order Verlet scheme for translation and a second order RATTLE scheme for rotation. Details can be found in [11]. The solid rigid body is supposed to be of general polygonal shape, and the computation of intersections in step (4) is straightforward.

Let the indices l , r , b and t denote the respective left, right, bottom and top faces of a cell \mathcal{C} . Following [8], the cut-cells computation in step (5) is written as follows in the 2D case:

- For each solid boundary \mathcal{F} , compute the quantity $\Delta w_{\mathcal{F}}^n$ swept by boundary \mathcal{F} , such that :

$$\sum_{\mathcal{F}} \Delta w_{\mathcal{F}}^n = \sum_{\mathcal{C}} \left(\alpha_{\mathcal{C}}^{n+1} - \alpha_{\mathcal{C}}^n \right) w_{\mathcal{C}}^n \quad (1)$$

where w denotes alternately density ρ , momentum $\rho \mathbf{u}$ or energy ρe . In practice, we compute $\Delta w_{\mathcal{F}}^n$ as the integral of w^n in the quadrangle defined by the positions of \mathcal{F} at time $n\Delta t$ and $(n+1)\Delta t$.

- Compute the numerical fluxes $F_{\mathcal{F}}^{\text{num}}$ at the solid boundary using p_x^{ref} and p_y^{ref} stored in the boundary \mathcal{F} . The fluxes are given by:

- If $w = \rho$, $F_{\mathcal{F}}^{\text{num}} = 0$
- If $w = \rho u$, $F_{\mathcal{F}}^{\text{num}} = \frac{1}{S_{\mathcal{F}}} \int_{\mathcal{F}} p_x^{\text{ref}} n_{\mathcal{F}}^x$

- If $w = \rho v$, $F_{\mathcal{F}}^{\text{num}} = \frac{1}{S_{\mathcal{F}}} \int_{\mathcal{F}} p_y^{\text{ref}} n_{\mathcal{F}}^y$
- If $w = \rho e$, $F_{\mathcal{F}}^{\text{num}} = \frac{1}{S_{\mathcal{F}}} \mathbf{V}_{\mathcal{F}}^{n+\frac{1}{2}} \cdot \int_{\mathcal{F}} \begin{pmatrix} p_x^{\text{ref}} n_{\mathcal{F}}^x \\ p_y^{\text{ref}} n_{\mathcal{F}}^y \end{pmatrix}$

- Compute $\Delta w_{\mathcal{C}}$ for each cell \mathcal{C} :

$$(1 - \alpha_{\mathcal{C}}^{n+1}) \Delta w_{\mathcal{C}} = \Delta t \left(\frac{1 - \kappa_{\mathcal{C}l}^{n+1}}{\Delta x} F_{\mathcal{C}l}^{\text{num}} - \frac{1 - \kappa_{\mathcal{C}r}^{n+1}}{\Delta x} F_{\mathcal{C}r}^{\text{num}} + \frac{1 - \kappa_{\mathcal{C}b}^{n+1}}{\Delta y} F_{\mathcal{C}b}^{\text{num}} - \frac{1 - \kappa_{\mathcal{C}t}^{n+1}}{\Delta y} F_{\mathcal{C}t}^{\text{num}} \right) + \sum_{\mathcal{F} \in \mathcal{C}} \frac{\Delta t S_{\mathcal{F}}}{\Delta x \Delta y} F_{\mathcal{F}}^{\text{num}} + \sum_{\mathcal{F} \in \mathcal{C}} \Delta w_{\mathcal{F}}^n \quad (2)$$

- Update the value of $w_{\mathcal{C}}$ in every cell :

$$w_{\mathcal{C}}^{n+1} = w_{\mathcal{C}}^n + \Delta w_{\mathcal{C}} \quad (3)$$

To ensure stability of the method for small cut-cells, we follow the procedure described in [8], with minor changes due to the impossibility to define a normal vector in a cell occupied by two boundaries. We define small cells as $\alpha_{\mathcal{C}} > 0.5$. If we mix cell \mathcal{C} with a target cell \mathcal{C}_t , so that the final value w of the two cells is equal, we have to exchange the quantities $M_{\mathcal{C}\mathcal{C}_t} = \frac{\alpha_{\mathcal{C}_t}}{\alpha_{\mathcal{C}} + \alpha_{\mathcal{C}_t}} (w_{\mathcal{C}_t} - w_{\mathcal{C}})$ and $M_{\mathcal{C}_t\mathcal{C}} = \frac{\alpha_{\mathcal{C}}}{\alpha_{\mathcal{C}} + \alpha_{\mathcal{C}_t}} (w_{\mathcal{C}} - w_{\mathcal{C}_t})$, and we easily check that $w_{\mathcal{C}} + M_{\mathcal{C}\mathcal{C}_t} = w_{\mathcal{C}_t} + M_{\mathcal{C}_t\mathcal{C}}$. In the 2D case, we have to make a choice for the target cell \mathcal{C}_t . We fix \mathcal{C}_t to be the fully-fluid cell ($\alpha_{\mathcal{C}_t} = 0$) nearest to cell \mathcal{C} , such that the path between the two cells does not cross a solid boundary. A recursive subroutine finds such a target cell after few iterations.

Let us note that the mixing procedure is entirely conservative, and ensures that the significant volume for a cell is consistent with the usual CFL condition based on standard cell size.

3 Theoretical Results

The following results were theoretically proven:

- **Mass, momentum and energy conservation:**

When there is no inflow into or outflow from the domain, conservation of fluid mass is ensured. For periodic boundary conditions, the momentum is exactly balanced between fluid and solid during each time-step. For periodic or reflecting boundary conditions, the energy received by the fluid from the solid is exactly balanced by the work of fluid pressure forces on the solid during each time-step.

- **Frame indifference:**

Let an arbitrary shaped rigid body moving at constant velocity and no rotation, be immersed in a uniform fluid flow at the same velocity. The uniform motion of both the solid and the fluid is preserved by the coupling algorithm.

- **Free slip along a straight boundary:**

A uniform flow parallel to a rigid semi-infinite half-plane is preserved by the coupling algorithm.

The last result shows that no numerical boundary layer or artificial boundary roughness appear at the solid boundary, even when it is not aligned with the Cartesian mesh.

4 Numerical Results

Due to space limitations, we only present here a moving boundary benchmark that was first proposed in [7] and also treated in [8]. A rigid cylinder of density 7.6 kg m^{-3} , initially resting on the lower wall of a two-dimensional channel filled with air at standard conditions, is driven and lifted upwards by a Mach 3 shock wave. The results obtained on a $1,600 \times 320$ grid are shown in Fig. 1. We observe good agreement with the results shown in [2, 8]. Small differences in the position of the shock waves can be noticed but no reference solution exists for this case. We also observe a strong vortex under the cylinder which is much weaker in [8]. This vortex seems to be associated with a Kelvin-Helmholtz instability originating at the contact discontinuity present below the cylinder.

In addition, when considering the final position of the center of mass of the cylinder, we observe a fast grid convergence of the method. The position we obtain on a 400×80 -grid is comparable to that obtained on a $1,600 \times 320$ -grid in [8]. With increasing resolution on grids 400×80 , 800×160 and $1,600 \times 320$ [8] gives

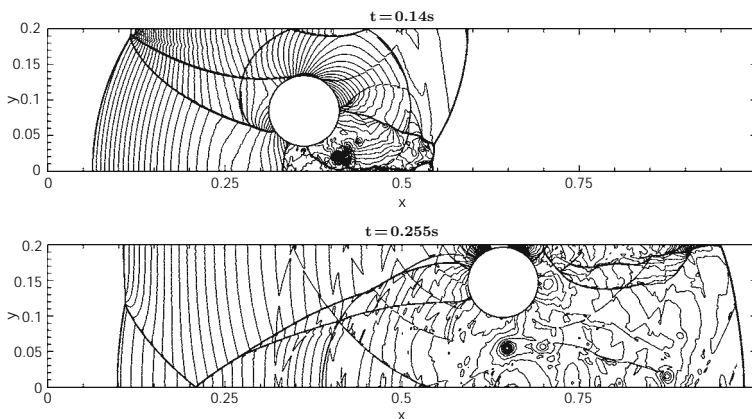


Fig. 1 60 contours of fluid pressure from 0 to 28 at different times, $\Delta x = \Delta y = 6.25 \times 10^{-4}$

positions (0.659, 0.132), (0.649, 0.145) and (0.641, 0.147); on the same grids, we obtain (0.64375, 0.1463), (0.64278, 0.1471) and (0.64253, 0.1471).

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